

APPENDIX C

ATLAS

### Structural Loads

Limit load factors for various mission events are given in Table C-1 for the commercial Atlas I, II, IIA, and IIAS launch vehicles. The load factors given are intended to provide a conservative design envelope for a typical spacecraft in the 1800 kg (4000 lb) to 3600 kg (8000 lb) weight class with first lateral modes above 10 Hz and first axial mode above 15 Hz. In addition, the center of gravity offset from the payload adapter interface is in the range of 89-152 cm (35-60 inches).

Gust/flight wind is a low frequency event (<12 Hz) that produces maximum loss of clearance between the spacecraft and payload fairing, and high loads near the base of the spacecraft primary structure. BECO/BPJ excites all frequencies (3 to 40 Hz) and produces the majority of the maximum loads throughout the spacecraft. MECO excites all frequencies and produces the highest tension (negative axial) loads and sometimes the maximum loads on secondary structure.

Table C-1  
ATLAS I, II, IIA & IIAS  
Limit Load Factor (G)  
at spacecraft C.G.

Event	Axial	Lateral
Launch		
ATLAS I, II, IIA	1.2 ± 1.2	± 1.0
ATLAS IIAS	1.3 ± 1.8	± 1.3
Flight Winds	2.2 ± 0.3	0.4 ± 1.2
BECO/BPJ		
(max axial)	5.2 ± 0.5	± 0.5
(max lateral)	2.5-1.0 ± 1.0	± 2.0
SECO*	2.0-0.0 ± 0.4	± 0.3
MECO*	4.0-0.0 ± 0.5	± 0.2
	0.0 ± 2.0	± 0.6

Note: Dynamic Uncertainty Factors (DUF's) are not accounted for in the above load factors.

\* Decaying to zero.

+ Is compression.

BECO = Booster Engine Cut-off  
BPJ = Booster Package Jettison  
SECO = Sustainer Engine Cut-off  
MECO = Main Engine Cut-off

### Acoustics

Qualification and acceptance acoustic test levels are given in Tables C-2 and C-3 for the ATLAS I, II, or IIA with 3.4 m (11-ft) and 4.3 m (14-ft) payload fairings respectively. The acoustic levels for the ATLAS IIAS are given in Table C-4.

For the 4.3 m (14-ft) payload fairing with acoustic blanket, special consideration should be given to components located within 76 cm (30-in.) of the payload fairing vents; the expected sound pressure level can be greater than the levels given in Tables C-3 and C-4 at higher frequencies. Table C-5 gives expected Sound pressure levels for components located 0.3 m (1 ft) from the vents. The 3.4 m (11-ft) payload fairing vents are fewer in number and located farther from the spacecraft envelope.

### Spacecraft Random Vibration

The maximum expected random vibration flight levels (limit levels) at the spacecraft interface are given in Table C-6.

### Sine Vibration

The maximum expected sine vibration levels given in the ATLAS user's guide are given in Table C-7.

### Mechanical Shock

Test levels representing typical spacecraft separation and payload nose fairing and insulation panel jettison are given in Tables C-8.

The maximum acceptable shock level at the equipment module interface for a customer-provided separation system is given in Figure C-1.

Table C-2  
 ATLAS I, II, & IIA  
 Acoustic Test Levels  
 Inside 3.4 m (11 ft) Payload Fairing  
 Assumes 50-60% Fill by Cross Section Area)

One-Third Octave Center Frequency (Hz)	Noise Level (dB) re: .00002 Pa			
	Without Acoustic Blankets		With Acoustic Blankets	
	Qualification	Acceptance	Qualification	Acceptance
25	121	118	121	118
32	123	120	123	120
40	124.5	121.5	124.5	121.5
50	126	123	126	123
63	128	125	128	125
80	129	126	129	126
100	130.5	127.5	130	127
125	132	129	131	128
160	132.5	129.5	131	128
200	133.5	130.5	131.5	128.5
250	134	131	131	128
315	133	130	129	126
400	132	129	127	124
500	131	128	125	122
630	129.5	126.5	123.5	120.5
800	127	124	121	118
1000	125	122	119	116
1250	122	119	116	113
1600	120	117	114	111
2000	119	116	113	110
2500	118.5	115.5	112.5	109.5
3150	118	115	112	109
4000	117.5	114.5	111.5	108.5
5000	117	114	111	108
6300	116.5	113.5	110.5	107.5
8000	116	113	110	107
10000	115.5	112.5	109.5	106.5
Overall	143	140	140	137

Table C-3  
 ATLAS I, II, & IIA  
 Acoustic Test Levels  
 Inside 4.3 m (14 ft) Payload Fairing  
 Assumes 50-60% Fill by Cross Section Area)

One-Third Octave Center Frequency (Hz)	Noise Level (dB) re: .00002 Pa			
	Without Acoustic Blankets		With Acoustic Blankets	
	Qualification	Acceptance	Qualification	Acceptance
25	116	113	116	113
32	120	117	120	117
40	123.5	120.5	124.5	121.5
50	124.5	121.5	125.5	122.5
63	125.5	122.5	127	124
80	126.5	123.5	128	125
100	127	124	129.5	126.5
125	128	125	131	128
160	128	125	132	129
200	128	125	133	130
250	127.5	124.5	132.5	129.5
315	127	124	131.5	128.5
400	126	123	131	128
500	123.5	120.5	130.5	127.5
630	121.5	118.5	129	126
800	119.5	116.5	126.5	123.5
1000	116.5	113.5	123.5	120.5
1250	114.5	111.5	121	118
1600	113	110	121	118
2000	113	110	122	119
2500	111	108	119.5	116.5
3150	111	108	118	115
4000	110.5	107.5	117	114
5000	110	107	116.5	113.5
6300	110.5	107.5	116	113
8000	112.5	109.5	116.5	113.5
10000	113.5	110.5	117.5	114.5
Overall	138	135	142	139

Table C-4  
 ATLAS IIAS  
 Acoustic Test Levels  
 Inside 4.3 m (14-ft) Payload Fairing  
 Assumes 50-60% Fill by Cross Section Area)

One-Third Octave Center Frequency (Hz)	Noise Level (dB) re: .00002 Pa			
	Without Acoustic Blankets		With Acoustic Blankets	
	Qualification	Acceptance	Qualification	Acceptance
25	117	114	117	114
32	121	118	121	118
40	124.5	121.5	125	122
50	125.5	122.5	126	123
63	127	124	127.5	124.5
80	127.5	124.5	129	126
100	128.5	125.5	130.5	127.5
125	129	126	132	129
160	129.5	126.5	133	130
200	130	127	134	131
250	129.5	126.5	133.5	130.5
315	129.5	126	133	130
400	128	125	133	130
500	126.5	123.5	133	130
630	125	122	131.5	128.5
800	122.5	119.5	130	127
1000	119.5	116.5	127	124
1250	117	114	125	122
1600	115	112	123.5	120.5
2000	115	112	122	119
2500	114	111	121	118
3150	113	110	120	117
4000	112	109	118.5	115.5
5000	111.5	108.5	117.5	114.5
6300	111	108	116.5	113.5
8000	112.5	109.5	117	114
10000	113.5	110.5	118	115
Overall	139	136	143	140

Table C-5  
Acoustic Levels 0.3 m (1 ft) from the Vents  
for the 4.3 m (14 ft) Payload Fairing

One-Third Octave Center Frequency (Hz)	Noise Level (dB) re: .00002 Pa	
	Qualification	Acceptance
1600	113.5	110.5
2000	115.5	112.5
2500	115.5	112.5
3150	114	111
4000	115	112
5000	116.5	113.5
6300	116.5	113.5
8000	117	114
10000	117.5	114.5

Table C-6  
ATLAS I, II, IIA, IIAS  
Spacecraft Random Vibration  
Limit Levels

Frequency (Hz)	ASD Level ( $G^2/Hz$ )
20	.00048
20-80	+9 dB/oct
80-200	.03
200-2000	-9 dB/oct
2000	.00003
Overall	2.7 $G_{rms}$

Table C-7  
 ATLAS I, II, IIA, & IIAS  
 Maximum Expected Spacecraft Interface  
 Sinusoidal Vibration Environment

Frequency (Hz)		Sine Vibration Level ( $G_{0-p}$ )
Thrust Axis	5-6.2	12.5-mm (0.5-in) DA 1.0
	6.2-100	
Lateral Axes	5-100	0.7

Table C-8  
 ATLAS I, II, IIA, & IIAS  
 Spacecraft Separation  
 Shock Response Spectrum  
 $Q=10$

Event	Frequency (Hz)	Shock Response Spectrum (G)	
		Qualification	Acceptance
Spacecraft Separation			
Type D Payload Adapter [1.65 m (66 in)]	100 100-800 800-3000	210 +7.1 dB/oct 4200	150 +7.1 dB/oct 3000
Type B & B1 Payload Adapter [1.18 m (47 in)]	100 100-1500 1500-3000	140 8.5 dB/oct 6300	100 8.5 dB/oct 4500
Type A & A1 Payload Adapter [0.92 m (37 in)]	100 100-1500 1500-3000	70 10 dB/oct 6300	50 10 dB/oct 4500
Payload Fairing and Insulation Panel Jettison	100 100-500 1500-2000	20 5.4 dB/oct 84	14 5.4 dB/oct 60

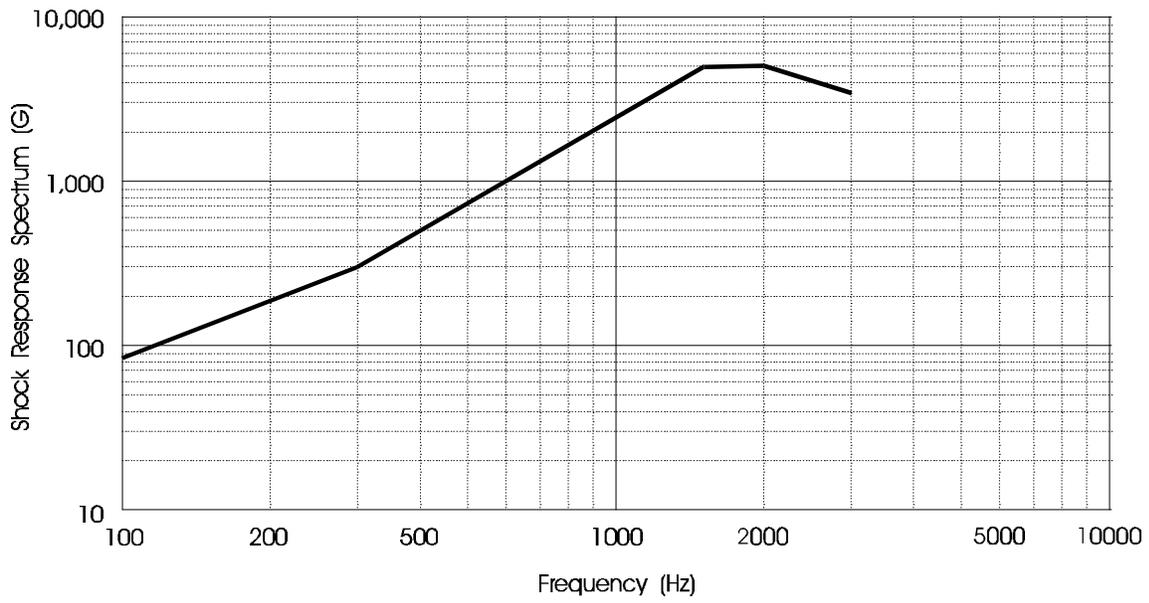


Figure C-1 Maximum Allowable Spacecraft-Produced Shock at Equipment Module Interface